

SUPPLEMENTAL TYPE CERTIFICATE

10071270

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

AIRFILM CAMERA SYSTEMS, INC

**6245 AERODROME WAY, HANGAR 2
GEORGETOWN CA 95634
USA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.R.008

Type Certificate Holder: AIRBUS HELICOPTERS

Type: AS350 / EC130

Model: 130 B4/ 130 T2

Original STC Number: FAA SR02695LA

Description of Design Change:

Installation of Airfilm Utility Mount

EASA Certification Basis:

The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

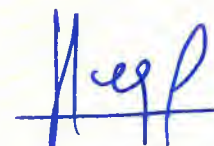
Associated Technical Documentation:

1. Airfilm Camera Systems documentation listed on AML No. SR02695LA, dated 11 April 2019, or later FAA approver revisions.
2. Instructions for Continued Airworthiness, which are found within Airfilm Camera Systems Report No. AFM-130-208, Rev. A, dated 19 February 2019, or later FAA accepted revisions.

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 17 October 2019



Fabrice LEGAY

Section Manager

Medium & Light Rotorcraft



Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -

