



## SUPPLEMENTAL TYPE CERTIFICATE

**10039428**

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EC) No. 1702/2003 to

### **AIRFILM CAMERA SYSTEMS, LLC**

**6245 AERODROME WAY, HANGAR 2  
GEORGETOWN CA 95634  
USA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

**Original Product TC Number : EASA.IM.R.506**

**TC Holder : BELL HELICOPTER TEXTRON CANADA**

**Model : 429**

**Original STC Number : FAA STC SR02377LA**

**Description of Design Change:**

Airfilm Nose Utility Mount

**EASA Certification Basis:**

The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

**For the European Aviation Safety Agency,**

**Date of issue: 03.05.2012**

  
**Massimo MAZZOLETTI**  
**Certification Manager**  
**Rotorcraft, Balloons, Airships**

**Note:**

The following numbers are listed on the certificate:  
EASA current Project Number: 0010015217-001

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**Associated Technical Documentation:**

- Master Drawing List AFM-NM429-10, Rev N/C, dated August 1, 2011
  - Installation Instructions AFM-NM-006, dated October 7, 2011
  - Instructions for Continued Airworthiness AFM-NM-008, dated August 1, 2011
  - Rotorcraft Flight Manual Supplement AFM-NM-005, dated October 17, 2011
- or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)

**Limitations:**

None

**Conditions:**

Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- end -

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